

**IN THE CLAIMS:**

Please enter the amended Claims as follows–

Claim 1. (amended) A combustor for a gas turbine engine, the combustor comprising an outer wall, an upstream end and a downstream end with fluid flow through said combustor progressing from said upstream end toward said downstream end and [a] an inner wall element comprising a main body member, a plurality of heat removal members on said main body member extending from said main body moving towards said outer wall, and at least one surface, the surface, in use, facing said downstream end relative to the general direction of fluid flow through the combustor and including a downstream facing surface of at least one of said of said heat removal members, wherein at least said downstream facing surface comprises a thermal barrier coating.

Claims 2-5 are canceled.

Claim 6. (amended) A combustor according to claim 1 wherein said heat removal members have a selected length and said thermal barrier coating extends substantially the whole length of said heat removal members.

Claim 7. (amended) A combustor according to claim 1 wherein the heat removal members [are upstanding] extend away from the main body member.

Claims 8-12 are canceled.

Claim 13. (original) A wall element according to claim 1 wherein the thermally resistant material is a thermal barrier coating.

Claim 14. (original) A wall element according to claim 13 wherein the thermal barrier coating is magnesium zirconate.

Claim 15. (original) A wall element according to claim 13 wherein the thermal barrier coating is yttria stabilized zirconia.

Claims 16-19 are canceled.